

## ***Interactive comment on “Radar observability of near-Earth objects using EISCAT 3D” by Daniel Kastinen et al.***

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Dear Peter, Thank you very much for the great feedback on the manuscript! We have treated the individual items below.

**For the case of mini-moon detection, no discussion is given of how to distinguish natural objects (which are comparatively rare) from much more common artificial objects in geocentric orbit. Can the authors comment on the challenges this would pose for radar detection of mini-moons?**

Generally space-debris are in a set of well defined orbital regions close to the Earth, orbits that are wildly different from those of minimoons and most NEOs. Therefore the range will be the primary identification as there is very little debris outside 3,000 km

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altitude, with the exception of GEO and GTO. As most mini-moon detections will occur much further away these are very likely not artificial object. However, the most secure method of identification is through the orbit of the detected object. It should be relatively straight forward to distinguish them from artificial objects if the orbit determination is of sufficient quality, especially if the radar astrometry allows for the estimation of the area-to-mass ratio which is typically used to discern between natural and artificial bodies (Jedicke et al. 2018). For the case of EISCAT 3D, describing how to do orbit determination of sufficient quality after initial discovery is a complete separate study in itself but it is definitely possible.

We have added the following on this to the discussion:

*It was shown in Kastinen et al. (2019) that E3D is expected to regularly detect hard target echoes from space debris and other artificial objects in Earth orbit. NEOs and minimoons need to be robustly separated from these artificial objects for discovery operations to be successful. Space debris is mainly confined to two regions: close to Earth ( $< 3 \times 10^3$  km altitude) or close to geostationary orbit ( $\sim 3.6 \times 10^4$  km altitude) (Krisiko, 2010; Flegel et al., 2009). Our results show that the typical minimoon or NEO detection will be made at altitudes larger than these regions and up to  $3.8 \times 10^5$  km altitude. Thus, range to the target can be used as an initial NEO and minimoon identification. If the orbit of the object can be determined, this would be a very reliable method of identification as NEOs and minimoons generally have vastly different orbits compared to space debris (Fedorets et al., 2017).*

**At the smallest sizes (sub-meter to decimeter) there is the question of what is the scientific value of radar detections? For such small objects can we estimate rotation rates or surface roughness or is the SNR too low? What can we learn from detecting such small objects which we would not learn but studying them ablate in the atmosphere?**

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First off, it is important to note that the populations that are mainly discussed here will not ablate in the atmosphere but rather pass the Earth by. E.g., only a very small fraction of the intermediate NEO population used to simulate the minimoon population will ablate in the atmosphere, as was shown in [Fedorets et. al 2017 Orbit and size distributions for asteroids temporarily captured by the Earth-Moon system]. With that in mind, the main contribution is that this is a population otherwise not observable! For the case of EISCAT 3D, it is not really the SNR that limits if a measurement of rotation rate or surface roughness can be made but rather the wavelength of the radar relative the target size. Therefore its not expected that it can derive such parameters for smaller objects. Rather, what phased array radars can do really well is possible mono-static orbit determination, good collection area by scanning, low restriction on observing conditions (clouds and sunlight do not impact measurements), the detection of small sizes and easier observational bias calculations. I.e. they produce "a lot" (this is relative of course) of high quality data that can be used for good population density estimates.

We have added the following to the discussion with respect to the above:

*The scientific gain from tracking operations at E3D can be summarized as: efficient, high quality orbit determination and, if the target is sufficiently larger than the wavelength, novel data on surface properties and rotation rates. There are currently not many methods that can discover smaller NEOs unless they collide with the Earths atmosphere, as shown by the low number of discovered minimoons. As such, the scientific gain from discovery operations is in essence the discovery itself, i.e. observation capability of a population otherwise not observable. If the objects are larger, they can be detected with higher probability using optical methods. In these cases radar observations are still valuable for the same reasons as tracking operations are.*

**The introduction is very complete, but the authors may want to consider**

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adding or discussing the one reference to the only published paper relating to attempted radar detection of decimeter sized NEOs/meteoroids: Kessler D. J., Landry P. M., Gabbard J. R., and Moran J. L. T. 1980. Ground radar detection of meteoroids in space. In In: **Solid particles in the solar system; Proceedings of the Symposium**, edited by Halliday I., and McIntosh B. A. IAU. p. 137.

The following has been added to the introduction:

*Smaller radars can be used for nearly continuous observing, and it is possible that they can even contribute to the discovery of NEOs. Kessler et al. (1980) presented an early attempt at discovering meteoroids outside of the Earths atmosphere using a space surveillance radar. However, the observation span was only 8 hours and the results were inconclusive, but 31 objects were identified as possible meteoroids. No followup studies were conducted.*

**Rotation rates of decimeter-sized meteoroids have been discussed and modelled in a few works such as: Capek D. 2014. Rotation of cometary meteoroids. *Astronomy and Ę Astrophysics* 568:1–8. Beech M., and Brown P. G. 2000. Fireball flickering: the case for indirect measurement of meteoroid rotation rates. *Planetary and Space Science* 48:925–932.**

These studies are very interesting! We have added the following to the discussion on rotation rates in the 2.3 Minimoons model section:

*In what follows we assume that the objects could have one of four different rotation rates: 1 000, 5 000, 10 000 or 86 400 revolutions per day. These values are also consistent with modeling of cometary meteoroids. For example, Capek (2014) studied the distribution of rotation rates of meteoroids ejected from 2P/Encke and found that objects with diameters between 1 to 10 centimeters have rotation rates approximately between 10 Hz to 0.1 Hz. There are also indirect observations of meteoroid rotation rates derived from optical meteor light-curve oscillations. Beech and Brown (2000) estimate Ę 20 Hz and less for objects larger than 10 centimeters in diameter.*

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**Line 245: transmitter bandwidth of < 5 Hz; transmitter bandwidth of < 30 Hz – I think the latter should be receiver bandwidth?**

Yes, that is right! There was also a missing MHz and It should read: "transmitter bandwidth of  $\leq$  5 MHz; receiver bandwidth of  $\leq$  30 MHz;". It has been changed to this.

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