



1 Vertical Ionosphere Delay Estimation using Zero Difference GPS

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Phase Observation

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8 Abstract

An apparent delay is occurred in GPS signal due to both refraction and diffraction 9 caused by the atmosphere. The second region of the atmosphere is the ionosphere. 10 The ionosphere is significantly related to GPS and the refraction it causes in GPS 11 signal is considered one of the main source of errors which must be eliminated to 12 determine accurate positions. GPS receiver networks have been used for monitoring 13 the ionosphere for a long time. 14 the ionospheric delay is the most predominant of all the error sources. This delay is 15 a function of the total electron content (TEC). Because of the dispersive nature of 16 the ionosphere, one can estimate the ionospheric delay using the dual frequency 17 GPS. 18 In the current research our primary goal is applying Precise Point Positioning (PPP) 19 observation for accurate ionosphere error modeling, by estimating Ionosphere delay 20 using carrier phase observations from dual frequency GPS receiver. The proposed 21 algorithm was written using MATLAB. 22

- The proposed Algorithm depends on the geometry-free carrier-phase observations
- 24 after detecting cycle slip to estimates the ionospheric delay using a spherical
- ionospheric shell model, in which the vertical delays are described by means of a
- 26 zenith delay at the station position and latitudinal and longitudinal gradients.
- 27 geometry-free carrier-phase observations were applied to avoid unwanted effects
- of pseudorange measurements, such as code multipath. The ionospheric
- estimation in this algorithm is performed by means of sequential least-
- 30 squares adjustment.
- Finally, an adaptable user interface MATLAB software are capable of estimating
- ionosphere delay, ambiguity term and ionosphere gradient accurately.





33 **1. Introduction**

- 34 During the transmission of GPS signals from satellite to receiver, the signals
- ³⁵ propagate through the ionosphere so that the ionospheric delay is closely associated
- with GPS and is considered one of the main sources of errors in point positioning
 using GPS techniques, on the other hand GPS can be used as a sensor of the
 ionosphere and investigate its characteristics because of the global system coverage
- and the availability of multiple frequency data.
- 40 In this paper we used GPS receiver as a sensor of the ionosphere. The ionosphere
- is a dispersive medium, which means that the delay depends on the frequency of
- 42 the signal. the first order effect of the ionosphere refraction could be eliminated
- 43 mathematically by means of a linear combination of the signals on the two
- 44 frequencies, because GPS signals are broadcast on more than one frequency. This
- 45 combination is widely called the iono-free combination (Leandro, 2009).
- 46 Various methods were devised to calculate the ionospheric delay. These methods
- were based on spherical harmonic expansions in the global or regional scale (e.g.
 Schaer, 1999, and Wielgosz et al., 2003a). Local methods were based on two-
- dimensional Taylor series expansions (e.g. Komjathy, 1997, Jakobsen et al. 2010,
- 50 Deng et al 2009, and Masaharu et al. 2013).
- 51 This paper is aimed to apply Precise Point Positioning (PPP) observation for
- ⁵² accurate ionosphere error modeling, using carrier phase measurements the
- 53 proposed algorithm was written using MATLAB.

2. Observations equations for carrier-phase measurements.

The observations of dual-frequency GPS receiver at any station consists of two codes and two carrier phase observations in RINEX format which were used for present model. The observations equations for carrier-phase measurements can be formulated as follows (Leandro, 2009; e.g. Sedeek et al., 2017):

59
$$\Phi = R + c(dT - dt) + T - I + \lambda N + pbr - pbs + hdr - hds + m + e$$
(1)

Where Φ , R, C, dT and dt, T, I, Y, N, λ , hdr and hds, pbr and pbs and m are the 60 carrier-phase measurements, in meter, the geometric distance between satellite and 61 receiver antennas, in meters, the speed of light, in meters per second, the receiver 62 and satellite clock errors, respectively, in seconds, the neutral troposphere delay, in 63 meters, the ionosphere delay, in meters, the carrier-phase integer ambiguity, the 64 carrier-phase wave length, in meters, the receiver and satellite carrier-phase 65 hardware delays, respectively, in metric units, the receiver and satellite carrier-66 phase initial phase bias, respectively, in metric units, the carrier-phase multipath, in 67 meters, respectively and *e* is the un-modeled errors of carrier-phase measurements, 68 in meters. 69





3. Ionospheric Delay Estimation by Geometry-Free Linear Combination of GPS Observables.

The geometry-free linear combination of GPS observations is classically used for ionospheric investigations. It can be obtained by subtracting simultaneous pseudo range (P1-P2 or C1-P2) or carrier phase observations (Φ 1- Φ 2). With this combination, the satellite – receiver geometrical range and all frequency independent biases are removed (Ciraolo et al., 2007). The ionospheric estimation is performed using the following model (Leandro,2009):

78
$$\phi_{GF} = \phi_{L1} - \phi_{L2} = (1 - \gamma) MF \left(I_{\nu,0} + \nabla_{\phi} (\phi_P - \phi_0) + \nabla_{\lambda} (\lambda_P - \lambda_0) \right) + Nb'_{gf}$$
 (2)

where ϕ_{GF} is the geometry-free carrier-phase observation in length units, MF is the 79 ionosphere mapping function, $I_{v,0}$ is the vertical ionospheric delay at the station 80 position, $\nabla \phi$ and $\nabla \lambda$ are latitudinal and longitudinal gradients, respectively, ϕ_P 81 and λ_P are the geodetic latitude and longitude of the ionospheric piercing point, ϕ_0 82 and λ_0 are the geodetic latitude and longitude of the station, y is the factor to convert 83 the ionospheric delay from L1 to L2 frequency, unitless and Nb'_{gf} is an ambiguity 84 parameter which includes the carrier-phase integer ambiguity plus a collection of 85 biases. The mapping function is based on a spherical ionospheric shell model as 86 shown in Figure 1, and is computed according to (Leandro, 2009): 87

88 MF =
$$\sqrt{1 - \left(\left(\frac{r}{(r+sh)}\right)\cos(e)\right)^2}$$



Figure (1): Elements of the ionospheric shell model (Leandro, 2009).

(3)



(7)



- where r is the mean radius of earth, *sh* is the ionospheric shell height (default value
- is 350 km), β is the satellite elevation angle at the shell height piercing point, and *e*
- is the elevation angle of satellite S from station O as seen in Figure (1).
- To compute elevation and azimuth angle for any satellite (*e*, *Azim*), the receiver
- position in Earth Centered Earth Fixed (ECEF) is converted to geodetic coordinate
- 95 (λ, φ, z) . Then, the satellite position coordinate (x_s, y_s, z_s) from ECEF at the specified
- 96 epoch is interpolated from the IGS final orbits. The interpolated satellite position is
- then transformed to a local coordinate frame, East, North, and Up (ENU) system.
- The transferred ENU is used to calculate elevation and azimuth angles as follows
- 99 (Dahiraj, 2013 and Sedeek et al, 2017):

$$100 \qquad e = tan^{-1} \left(\frac{x_U}{\sqrt{x_N^2 + x_E^2}} \right) \tag{4}$$

$$101 \quad Azim = tan^{-1} \left(\frac{X_E}{X_N} \right) \tag{5}$$

Where *e*, A_{zim} are the elevation and azimuth angle of satellite at the receiver station respectively and X_E , X_N , X_U are the satellite position in local coordinate frame.

Usually, the ionosphere is assumed to be concentrated on a spherical shell located
at altitude (nominally taken as 350 km above Earth's surface. Ionospheric Pierce
Point is the intersection point between the satellite receiver line-of-sight, and the
ionosphere shell as shown in Figure (1).

108 IPP location can be computed by providing reference station coordinate (ϕ_0 , λ_0), 109 then the geographic latitude and longitude of IPP can be computed according to 110 elevation and azimuth angle of satellite (Dahiraj, 2013). The offset angel between 111 the IPP and the receiver (Ψ) is defined as the offset between the IPP and the user's 112 receiver. The elevation angle β and the offset angel between the IPP and the receiver 113 Ψ are computed as follow (El-Gizawy, 2003):

114
$$\beta = \cos^{-1}\left(\left(\frac{r}{(r+sh)}\right)\cos(e)\right)$$
(6)

115
$$\Psi = \beta - e = \cos^{-1}\left(\left(\frac{r}{(r+sh)}\right)\cos(e)\right) - e$$

116 Where r and sh are the mean radius of the spherical Earth and the height of IPP,

- 117 respectively. Given the user's receiver coordinates (ϕ_0 , λ_0), and the offset angle Ψ ,
- the pierce point coordinates $(\phi_{IPP}, \lambda_{IPP})$ are then derived by the following
- 119 expressions (El-Gizawy, 2003):



(10)



120
$$\phi_{IPP} = (\phi_r + \Psi \cos(\text{Azim}))$$
 (8)
121 $\lambda_{IPP} = \left(\lambda_r + \frac{\Psi \sin(\text{Azim})}{\cos(\phi_{IPP})}\right)$ (9)

The ionospheric estimation is performed by means of sequential least-squares adjustment, where the parameters are the ionospheric model elements (vertical delay

and gradients) and the ambiguities as follows:

125
$$L=AX$$

Where: L is the vector of observations, A is the design matrix, X is unknown parameters vector, and P is weight matrix of observations.

128
$$X = (A_1^T \cdot P_1 \cdot A_1 + A_2^T \cdot P_2 \cdot A_2)^{-1} (A_1^T \cdot P_1 \cdot L_1 + A_2^T \cdot P_2 \cdot L_2)$$
(11)

- By using this system of equations, vertical ionospheric delay, latitudinal and
- longitudinal gradients values at the station position are computed on an epoch byepoch basis.

132 4. Results and Discussions

- 133 In the present contribution, to evaluate the performance of the proposed model,
- numerical case-studies were performed on ten IGS stations. These stations are
- shown in Figure (2).

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- 138 The Ionosphere delay is estimated for observations of Doy 3, 2018 for these
- stations and the results were compared with the results of the online version of the
- 140 GPS Analysis and Positioning Software (GAPS) as shown in the following
- 141 figures:



143 Figure (3): Vertical Ionosphere delay of ALGO station estimated by the VIDE program and GAPS of DOY 3, 2018.



145 Figure (4): Vertical Ionosphere delay of CEBR station estimated by the VIDE program and GAPS of DOY 3, 2018.







































Figure (11): Vertical Ionosphere delay of METS station estimated by VIDE program and GAPS of DOY 3, 2018.





162 Previous figures show a comparison of the ionospheric delays computed with the

proposed code and GAPS. This comparison shows how much the accuracy of this

study is good in terms of agreement of solutions provided by GAPS.





- **Table (1):** The average Ionospheric Delay of each station of DOY 3, 2018 using the
- 166 Proposed code and GAPS.

	Average Ionospheric Delay (m)			Average Ionospheric Delay (m)	
Station	Proposed Code	GAPS	Station	Proposed Code	GAPS
ALGO	1.6205	0.9996	HUEG	0.9790	0.9328
CEBER	1.1203	0.7948	MADR	1.6838	0.8126
FRDN	1.3139	0.9387	MAT1	0.8189	0.9771
HERS	0.6594	1.0255	METS	0.4961	0.5106
HRAO	0.9681	1.2758	NRC1	1.2463	1.0848

167 **5.CONCLUSIONS**

We have overviewed an algorithm which can be used to estimate ionospheric delays 168 of GPS observations using single GPS receiver using a spherical ionospheric shell 169 model. This Algorithm depends on the geometry-free carrier-phase observations 170 after detecting cycle slip. The ionospheric estimation in this algorithm is performed 171 172 by means of Sequential least-squares adjustment. This study is performed on ten IGS stations. Previous figures and table (1) show an agreement of the proposed code 173 results and values provided by GAPS. This procedure may be better than GAPS 174 because it can estimate the ionospheric delays each thirty seconds whereas GAPS 175 estimate the ionospheric delays each ten minutes. 176

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