

Electric field measurements in a NLC/PMSE region during the MASS/ECOMA campaign

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Abstract. We present results of electric field measurements made during the MASS rocket campaign in Andøya, Norway into noctilucent clouds (NLC) and polar mesospheric summer echoes (PMSE) on 3 August 2007. The instrument used high input-impedance preamps to measure vertical and horizontal electric fields. No large-amplitude geophysical electric fields were detected in the cloud layers, but significant levels of electric field fluctuations were measured. Within the cloud layer, the probe potentials relative to the rocket skin were driven negative by incident heavy charged aerosols. The amplitude of spikes caused by probe shadowing were also larger in the NLC/PMSE region. We describe a method for calculating positive ion conductivities using these shadowing spike amplitudes and the density of heavy charged aerosols.

Keywords. Atmospheric composition and structure (Aerosols and particles) – Ionosphere (Electric fields and currents) – Meteorology and atmospheric dynamics (Atmospheric electricity)

1 Introduction

Noctilucent clouds (NLCs) are high altitude (82 km) ice clouds found at the polar summer mesopause. NLC have been observed by eye and camera since 1885, though modern measurements are most often done using lidar. Related to NLC are polar mesospheric summer echoes (PMSE), which are unusually strong radar echoes. They are also found at the polar summer mesopause, and are often co-located with NLC (Rapp and Lübken, 2004). PMSE are thought to be caused by subvisible ice particles, which then grow and sediment until they become visible as NLC (Cho and Röttger,

1997; Rapp and Lübken, 2004). Source mechanisms for both are still not completely understood, however. PMSE were discovered with the introduction of 50 MHz radar at Poker Flat, Alaska (Ecklund and Balsley, 1981). They are caused by fluctuations in the free electron density on length scales on the order of the radar half-wavelength. Such fluctuations are within the viscous subrange, and therefore were not expected to last long – on the order of hours – at that scale and altitude (Cho and Kelley, 1993). These density fluctuations are thought to be maintained by large, negatively charged aerosols which have much lower diffusivity than the ambient plasma (Rapp and Lübken, 2004).

Though there are models for NLC and PMSE formation which largely agree with ground-based observations (Lübken et al., 2008; Rapp and Lübken, 2004; Rapp and Thomas, 2006), there are less data available for the comparison of in situ measurements to appropriate models. There are few direct observations of turbulence and its effects on the electrical structure of NLC (Lübken et al., 2002). The electron density fluctuations that reflect radar should create measurable electric field fluctuations (Lie-Svendsen et al., 2003a,b; Robertson, 2007). Reduced conductivity due to electron scavenging by large aerosols can create large DC electric fields within NLC (Holzworth and Goldberg, 2004). Additionally, large voltage fluctuations were seen in the aft probes of both DROPPS rocket flights. Holzworth et al. (2001) and Sternovsky et al. (2004) showed that this was due to the probes rotating into and out of the rocket wake. A handful of rocket campaigns have flown electric field sensors to investigate these phenomena (Goldberg et al., 1993; Zadorozhny et al., 1993; Holzworth et al., 2001; Pfaff et al., 2001; Blix et al., 2003; Strelnikov et al., 2006), but the data has, thus far, been too variable to draw generally applicable conclusions.

The MASS/ECOMA rocket campaign was directed at detecting NLC particles and meteoritic condensation nuclei. It flew a large set of instruments, supported by the ALWIN radar (Latteck et al., 1999) and ALOMAR RMR lidar (von



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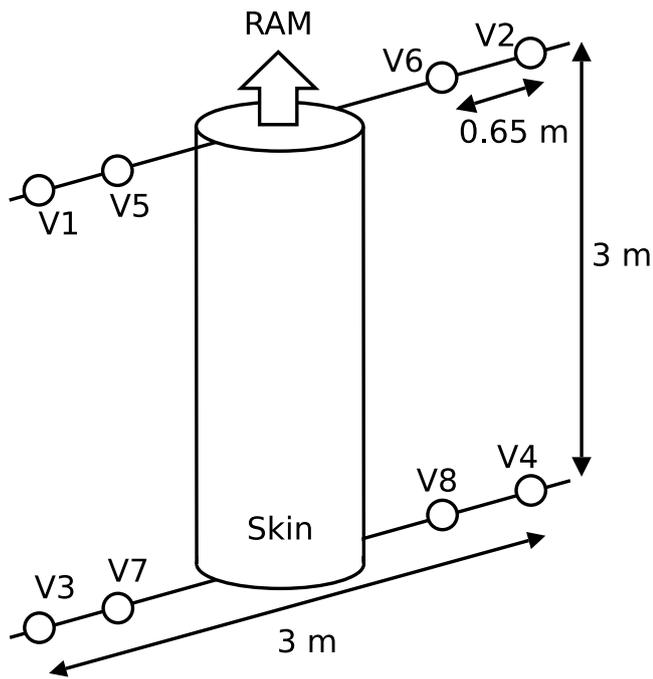


Fig. 1. MASS electric field probe arrangement. There were four booms with two probes each; two booms on the forward deck plate and two mounted on the aft deck plate. The 3'' spherical probes were coated with graphite paint. Voltage differences between pairs of probes, as well as the probe voltage relative to the rocket skin were telemetered.

Zahn et al., 2000), into NLC conditions to make a more comprehensive study of the phenomenon (Rapp et al., 2009; Robertson et al., 2009). In addition to allowing correlation studies between the ion densities and electric field within the cloud layer, the MASS and ECOMA rockets add to the variety of conditions into which these instruments have flown. This diversity of environmental conditions is essential for testing theoretical models throughout the parameter space. The two MASS payloads carried identical hardware (Robertson et al., 2009). The first flight went through moderate NLC and PMSE events, as observed by the ALOMAR lidar and ALWIN radar at Andøya rocket range, respectively; the second went through a somewhat weaker event. This paper will address the electric field fluctuations and conductivity measured during the first MASS flight.

We show here that the probes measured the expected $\mathbf{v} \times \mathbf{B}$ induced electric fields and spikes from shadowing of the probes by the rocket body. In addition, the probes moved about 1 V negative relative to the rocket body in the NLC region, possibly due to impacts of negatively charged aerosols. The probes did not detect significant (> 10 mV/m) DC fields and no wake effects were seen. We show that this small DC field perturbation is similar to that seen by the DROPPS 1 flight Holzworth et al. (2001). The electric field spectrum

was also found to be similar to the DROPPS 1 spectra from Pfaff et al. (2001), who suggested the turbulence was related to the strong PMSE radar reflections caused by structure at the Bragg scattering length. We also present a measurement of the upper bound of the conductivity of positive ions within the cloud layer. The measurement relies on the bottom boundary of the cloud layer being very sharp, and so the analysis cannot be extended to other datasets. However, the method works best within the cloud layer, which is a region that presents some difficulty for more conventional methods when large aerosols are present. We will report results on the order of 1×10^{-8} S/m which are consistent with measurements made by Croskey et al. (2001) in similar conditions.

2 Instrumentation

Figure 1 shows the configuration of the electric field probes on both MASS payloads. The presence of two probes on each boom allows for several different probe separation pairs. The booms and probes were mechanically identical to those flown on the DROPPS flights (Holzworth et al., 2001). The electronics, though not identical to DROPPS, were similar. They consisted of a high-impedance preamp in each spherical probe, followed by an amplifier module. Telemetry channels included voltage differences between pairs of probes as well as between the probes and the rocket skin. All channels were sampled with 12-bit accuracy, but sampling rates varied from 2 ksps to 33 ksps and gain varied from 0.25 to 10. Spatial resolution depends on the rocket velocity – about 1060 m/s – and was roughly 1 m–6 cm, depending on the sampling rate. Noise due to the electronics was only important in the least significant bit (0.6–23 mV, depending on the channel's amplification). This technique has been used many times in ionospheric rocket measurements (Fahleson et al., 1970; Pfaff et al., 1984; Maynard, 1998; Holzworth et al., 2001).

3 Data

3.1 Overview

The MASS 1 flight (NASA designation 41.069) launched on 3 August 2007 at 22:51 UTC. The solar zenith angle was about 93° , so the rocket was sunlit at the cloud altitude. Figure 2 shows the PMSE observed by the ALWIN radar. There is a double-peak structure, with peaks at 83 km and 88 km. There were low-altitude clouds present at launch, and so we do not have lidar data of the launch conditions. However, NLC were detected by ALOMAR during breaks in the clouds before and after the launch (Baumgarten et al., 2009).

Comparison of the measured electric field with the expected induced $\mathbf{v} \times \mathbf{B}$ electric field (Fig. 3) shows that these channels worked as expected. The induced $\mathbf{v} \times \mathbf{B}$ field was calculated and removed in subsequent plots. The field E_{12}

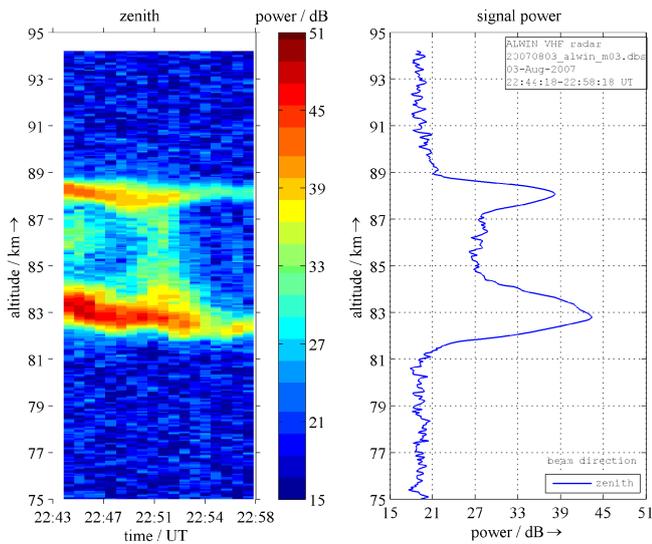


Fig. 2. ALWIN radar data for 3 August 2007 flight. T-0 was 22:51 UTC. Data and plot from Ralph Latteck at the Leibniz-Institut für Atmosphären Physik.

– which points from V_1 to V_2 – is given by $(V_1 - V_2)/d_{12}$, where V_1 is the voltage on probe V_1 , V_2 is the voltage on probe V_2 , and d_{12} is the distance between probes V_1 and V_2 . Figure 3 shows that E_{12} , E_{15} , E_{56} , and E_{34} all measure the same DC field – seen in the rotating rocket frame – despite having different probe separation distances. Spikes that are seen in all channels (for example, at 89.5 km, 90 km, 92.3 km, and 93.2 km) are caused by vibrations and/or gas releases from the attitude control system (ACS) valves. Small periodic deviations in the E_{15} signal are possibly due to wake effects. Due to this rocket going slightly faster than expected, the booms were deployed closer to the bottom of the cloud than expected and vibrations in the booms did not dampen out quickly enough to observe the probes' performance well below the NLC layer on the upleg portion. However, vibrations are absent within the cloud layer, and the probes behaved similarly until increasing air resistance broke the booms off around 65 km on the downleg. Thus, the probes accurately measured DC electric fields both above the cloud, and – on the downleg – below the cloud as well.

An overview of the aerosols and fields during the cloud passage is shown in Fig. 4. Figure 4a shows the charge density of the largest aerosols (>3 nm radii) measured by the MASS instrument, as a function of altitude (Robertson et al., 2009). Figure 4b shows the rocket skin voltage with respect to two sets of probes. The solid line is with respect to the average of the forward probes V_1 and V_2 – that is, $(V_{S1} + V_{S2})/2$ – and the dashed line is with respect to the average of the aft probes V_3 and V_4 . Large spikes in the $V_{S3} + V_{S4}$ data at 84, 84.5, and 87 km seen in Fig. 4b are due to ACS valve firings. The onset of the cloud layer at 83 km

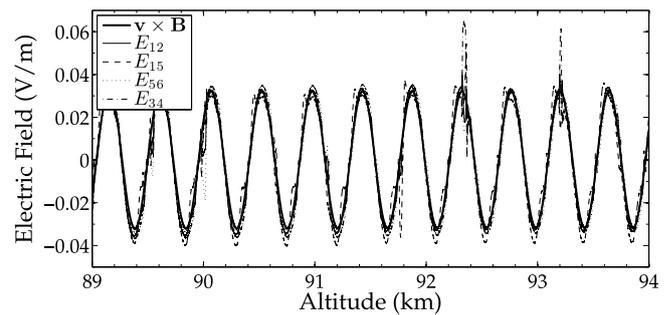


Fig. 3. Electric fields measured by several pairs of probes over a 5 km region above the NLC layer. These are compared to the expected induced $v \times B$ electric field, calculated using the measured v and B . The 2 Hz oscillation is due to rotation of the rocket. This shows that different pairs of probes, with different probe separations, in the front and aft of the rocket all measure the same expected DC electric field above the NLC event.

is very abrupt in the upper two panels, but the instruments resolve this rapid change with many data points during the transition. Notice that there are no large oscillations in the aft-probe data ($V_{S3} + V_{S4}$) as seen in DROPPS (Holzworth et al., 2001; Sternovsky et al., 2004). This may be due to the wake being smaller in this flight, or perhaps it was more symmetrically distributed about the rocket. Some differences between the forward and aft probes can be seen near 83 and 88 km, and are probably due to rocket wake effects. However, these differences are significantly smaller than those modeled in Sternovsky et al. (2004).

Figure 4c shows E_{12} , the forward electric field in the cloud layer. The $v \times B$ electric field associated with the rocket velocity across the Earth's magnetic field has been subtracted. The periodic sharp spikes are due to the probes rotating into the shadow of the rocket. The larger fluctuations at low altitudes actually begin at lower altitudes than what is plotted, and include what looks like a smaller patch of particles below the main NLC event (Robertson et al., 2009). Data below 82 km is rather noisy, however – particularly in the forward channels – due to boom deployment being slightly higher in altitude than expected. Large fluctuations at the lower altitudes (83–85 km) of the cloud layer give way to smaller, higher frequency fluctuations in the middle altitudes (85–87 km). The top of the cloud layer shows weaker fluctuations.

Figure 4c also shows that the unperturbed plasma in front of the rocket ram did not record large DC fields in the cloud – in contrast to the strong NLC with no PMSE case in DROPPS 2 presented by Holzworth and Goldberg (2004). This result calls into question theories which require large DC electric fields for the formation or maintenance of NLC conditions. The model presented by D'Angelo (2005), for example, requires electric fields on the order of a few tens of mV/m for reasonable aerosol growth rates.

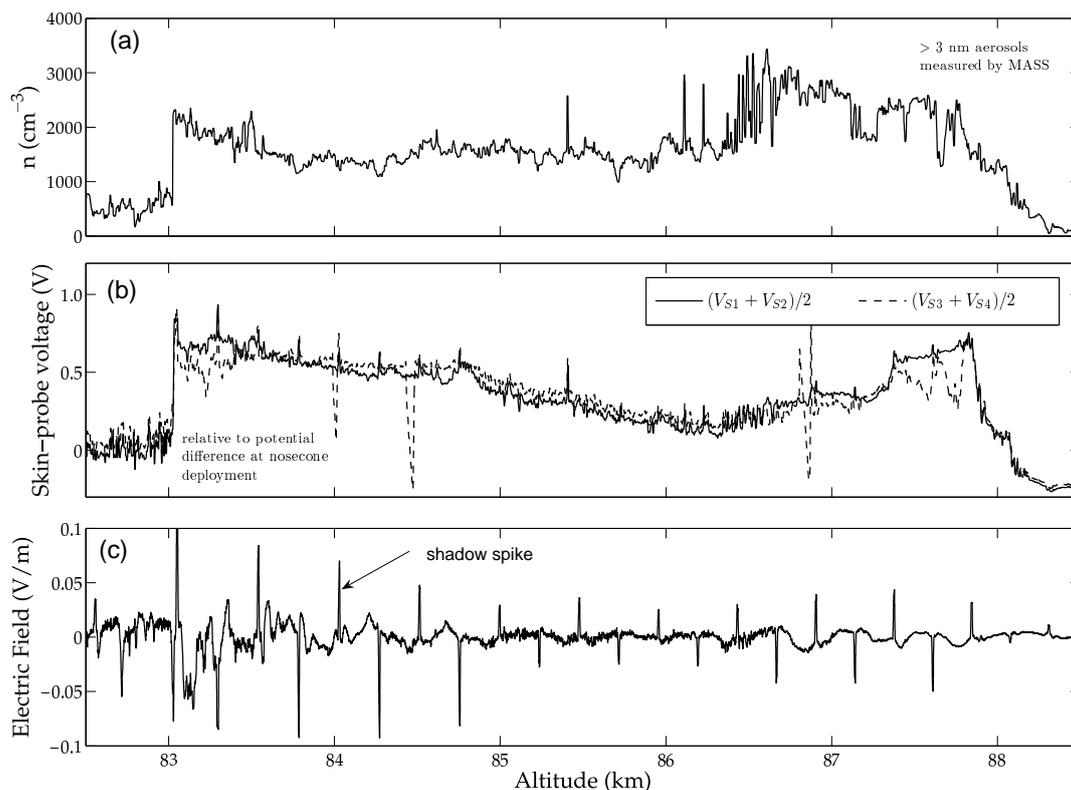


Fig. 4. An overview of the cloud layer. **(a)** Number densities of the heaviest aerosols measured by the MASS instrument, as a function of altitude (Robertson et al., 2009). **(b)** Change in the rocket-skin potential, with respect to the probe potential below the PMSE event, as a function of altitude. The solid line is with respect to the average of probes V1 and V2, and the dashed line is with respect to the average of probes V3 and V4. Notice how the profile follows the profile of the large negative aerosols. Large spikes in the $V_{S3}+V_{S4}$ data at 84, 84.5, and 87 km are due to ACS valve firings. **(c)** The E_{12} field, measured by probes V1 and V2 – that is, the field perpendicular to the rocket velocity. The $\mathbf{v} \times \mathbf{B}$ electric field associated with the rocket velocity across the Earth’s magnetic field has been subtracted. The sharp, periodic spikes are shadowing spikes, caused by the probe moving into the shadow of the rocket.

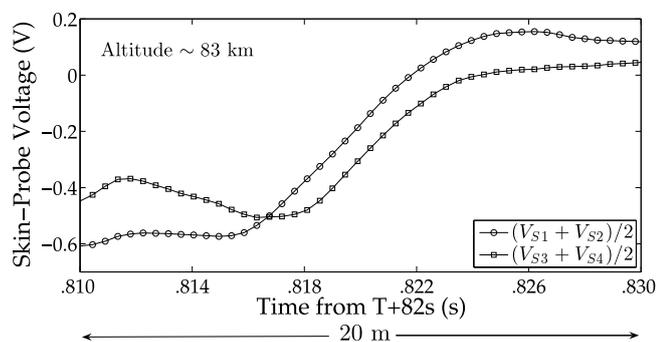


Fig. 5. A high time-resolution view of the onset of the NLC layer in Fig. 4b. The transition takes less than 10 ms, over much of which the potential varies linearly with time/altitude. There is a delay between the forward and aft channels of about 2.5 ms, which is roughly the time it takes for the aft probes to “catch up” to where the forward probes were (at the rocket velocity of about 1 km/s).

Comparing 4a and 4b, one can see that the shapes of the two curves are similar. There is a sharp ledge at 83 km, followed by a slow decrease, though it remains elevated relative to the value below the cloud. It starts increasing again above 86 km, and then drops back down by 88 km. The amplitude of the shadowing spikes also follows this general profile. This suggests that the skin-to-probe voltage and shadowing spike amplitudes are driven by the density of large charged aerosols.

3.2 Rapid potential changes

Figure 5 is an expanded view of Fig. 4b, which shows the onset of the bottom of the NLC layer at 83 km. The transition takes less than 10 ms (< 10 m), and over much of that time the voltage of the probes increases linearly with time/altitude, which suggests that a current to the probe has suddenly been “switched on”. In Fig. 5, one can see a delay between the forward and aft channels of about 2.5 ms, which is roughly the time it takes for the aft probes to “catch up” to where the

forward probes were (at the rocket velocity of about 1 km/s). These two observations suggest that the onset of the NLC layer is very sharp: less than the 3 m vertical probe separation. If it were not sharp, or if the feature was due to a shift in the rocket-skin potential, there would not be a delay in the two signals. If the change in the skin-to-probe voltage is due primarily to changes in the probes, then it means the probes are becoming charged negative within the NLC layer. The similarity between the shape of the plot showing the probe voltage versus altitude and the density of large negatively charged aerosols (Figs. 4a and 4b) supports this interpretation. This is not unexpected, since a much greater fraction of the probes' surface area was exposed to the aerosols, as compared to the rocket body, which was aligned along the ram direction.

3.3 Probe shadowing

Figure 4c shows E_{12} versus altitude, with the $\mathbf{v} \times \mathbf{B}$ electric field subtracted, as discussed above. The shadowing spikes occur when the probes enter the rocket's shadow. Within the rocket's shadow the photocurrent stops, which causes the shadowed probe to become more negative. The profile of the spike amplitudes has a similar shape to both the heavy charged aerosol densities (Fig. 4a) and the skin-to-probe voltage (Fig. 4b). That is, the spikes suddenly increase in amplitude around 83 km, then reduce significantly in amplitude near 86 km, grow again above 86 km, and are dramatically reduced above 88 km.

Probe voltages are determined by current balance. Over periods longer than the local relaxation time, the probes should be in current balance. The net current to the probes is zero, so that

$$\sum I = I_\gamma + I_a + I_e + I_i = 0, \quad (1)$$

where I_γ is the photocurrent, I_a is the current due to incident aerosols, I_e is the electron thermal current, and I_i is the ion current (Maynard, 1998). The current drawn by the electronics is negligible.

The electron current is non-linear with voltage, but through much of the cloud layer, the ion current can be approximated as linear as in the IV curve seen by a low voltage Gardien condenser in an NLC (Croskey et al., 2001). This is especially true if we consider only the differential conductivity, $\sigma = \Delta I / \Delta V$, as a perturbation of the sunlit state. In the following, we will assume I_i is approximately linear with probe voltage. Thus, ΔI_i is approximately described by Ohm's law: $J = \sigma E$. Substituting $J = I / (4\pi r^2)$ and $E = -V/r$ for conducting spheres – where r is radius of the spheres – we have

$$\Delta I_i = -4\pi r \sigma V_{\text{probe}} = -\Delta(I_\gamma + I_a + I_e). \quad (2)$$

Below the NLC layer, the photocurrent is dominant, and we expect the probe voltage to be positive with respect to the lo-

cal space potential. Above the NLC layer, the electron thermal current is dominant. In both cases, the dominant voltage-dependent current to the probes is carried by electrons. As such, the probe potential should be close to the local space potential, since the electrons have a high conductivity. This is consistent with data above the cloud layer (Fig. 4c, above 88 km).

Outside of the cloud, I_a is negligible. Within the cloud layer, however, the charged aerosol density – and I_a – is large enough that it cannot be ignored. When the probe is in the rocket shadow, the photocurrent vanishes and the net voltage-independent current is negative. The ambient electrons are cold – kT/q_e is about 12 mV – so the electron thermal current is negligible when the probe potential is less than the local space potential. Thus, during the shadowing spikes the return current is carried primarily by positive ions. Since the conductivity of ions is so much lower than for electrons, the shadowing spike amplitudes are much larger within the cloud. This also allows us to use the shadowing spike amplitude to approximate the potential difference due to the negative aerosol current.

An alternative analysis is to treat I_i , I_e , I_a , and I_γ as known (or calculable via a model), and to solve for the electron density that yields the correct shadowing spike amplitudes. In such a model, however, small electron densities are required for large shadowing spike amplitudes (and vice versa). PMSE peaks were measured at 83 km and 88 km, where the shadowing spikes were largest. In a double-layer PMSE event, one expects an electron density profile which has a minimum between the two peaks, where the radar reflection is weaker, instead of 2 minima co-located with the peaks. We nonetheless applied such a model, which calculated an electron density of 12 cm^{-3} at 83 km. It is unlikely that such a low electron density could support the PMSE measured there.

With that in mind, we continue trying to estimate the positive ion conductivity. It is important to know when the probes are in current balance. Figure 6 is a high resolution comparison of a tip-to-tip probe pair, V_{12} , and a same-boom probe pair, V_{15} , which shows that the probes reach current balance. First, probe V5 enters the shadow, which lowers its potential and increases the V_{15} potential difference. Then, probe V1 enters the shadow, which lowers its potential and decreases both V_{15} and V_{12} . Notice that V_{15} nearly returns to the same value it had before V5 entered the shadow. Afterwards, probes V1 and then V5 leave the shadow, reversing the voltage shifts.

Below about 84.5 km, however, the conductivity of the atmosphere is low enough that the probes do not quite reach equilibrium. That is to say, the local relaxation time is on the order of or longer than the duration of the shadowing spikes. Using this method below about 84.5 km overestimates conductivity. We use the slope in Fig. 5 to calculate the current

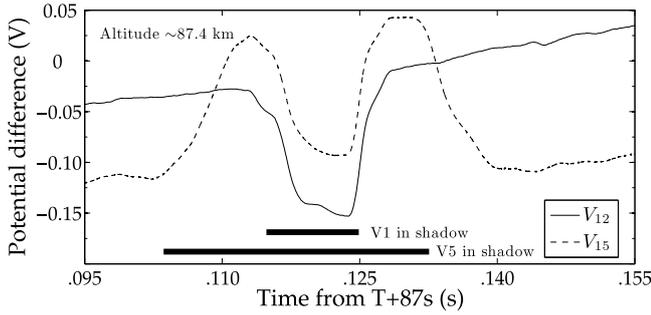


Fig. 6. High time resolution comparison of V_{12} and V_{15} at a shadowing spike around 87 km. Probe V2 is in sunlight the entire time. V5 goes into the rocket shadow first – raising the V_{15} potential difference – followed by V1. The probes are nearly settled in voltage, which shows that the current balance is attained within the rocket shadow. Below about 84.5 km, the conductivity is too low and the probes do not reach equilibrium, although the shadowing spike amplitudes are larger.

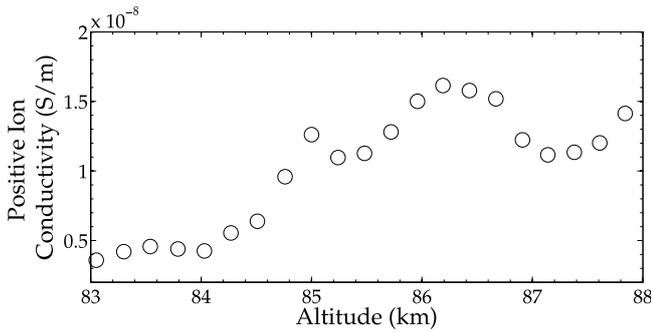


Fig. 7. Positive ion conductivity calculated from the shadowing spike amplitudes and aerosol densities. Points below about 84.5 km over-estimate the conductivity, since the shadowing spikes do not quite reach equilibrium within the rocket shadow. If there were other significant return currents present, they would inflate the calculated conductivity values. As such, this reflects the maximum value one could expect the ion conductivity to have.

to the probe via

$$I = C \Delta V / \Delta t, \tag{3}$$

and compare that to the number of particles measured by MASS at that altitude (Fig. 4a) to get an effective “efficiency” of charge collection,

$$\varepsilon = I_a / n_a(83 \text{ km}). \tag{4}$$

We then extrapolate to estimate I_a at other altitudes using Fig. 4a: $I_a \approx \varepsilon n_a$. This estimation, with shadowing spike amplitudes from Fig. 4c, yields an estimated positive ion conductivity,

$$\sigma = \varepsilon n_a / 4\pi r V_{\text{probe}}, \tag{5}$$

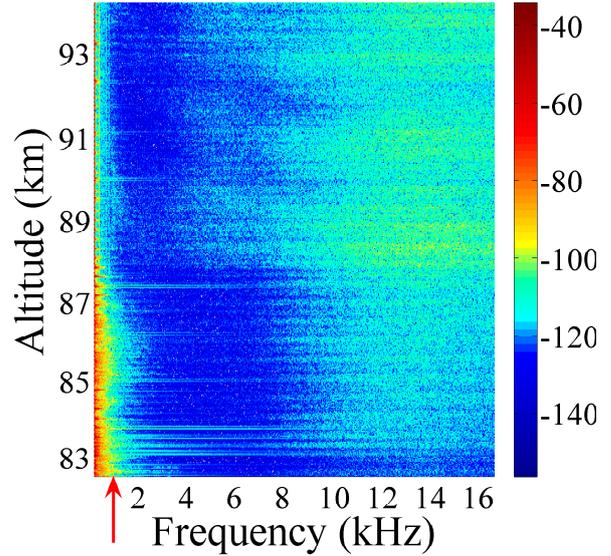


Fig. 8. Spectrogram of E_{12} around the PMSE layer. The layer can be seen around 83–88 km. The high-frequency fuzz is probably auroral hiss, which shows that the instrument accurately measured high frequency signals.

where the shadowing spike amplitudes are used for V_{probe} . Substituting $C = 4\pi\epsilon_0 \left(\frac{1}{r} - \frac{1}{r + \lambda_D}\right)^{-1}$ and Eq. (4) into Eq. (3) (where λ_D is the Debye length), this becomes

$$\sigma = \frac{\Delta V}{\Delta t} \frac{n_a}{n_a(83 \text{ km})} \frac{\epsilon_0}{V_{\text{probe}}} \left(1 - \frac{r}{r + \lambda_D}\right)^{-1}. \tag{6}$$

Figure 7 shows the results of this calculation. The overall values are within a factor of 2 of those measured by Croskey et al. (2001) on DROPPS 1. Note that below about 84 km, the values level off, as is expected, given that the probes did not reach equilibrium there. Also, if there were other significant return currents present, they would inflate the calculated conductivity values. As such, Fig. 7 reflects the maximum value one could expect the ion conductivity to have.

3.4 AC fluctuations

Two of the channels (E_{12} and E_{15}) used a sampling frequency of 33 kps, which allows for analysis of AC fields. Figure 8 shows a spectrogram of E_{12} . Ground-based instruments measured PMSE peaks at 83 and 88 km, and an NLC between 82–84 km, with a peak at 82.8 km (Robertson et al., 2009). The strong signals below 3 kHz in the 83–88 km altitude range (as denoted by the red arrow in Fig. 8) coincide with the measured PMSE, with the strongest signals occurring between the two PMSE peaks. It is not readily apparent why the strongest signals are between the peaks, as opposed to being coincident with the PMSE peaks, and may require more analysis.

Other channels look very similar. One might expect to see a difference in spectra between spin-plane channels and rocket-axis channels, since PMSE exhibit strong aspect sensitivity (Czechowsky et al., 1988). However, no easily identifiable differences were seen. This is not entirely unexpected, since the rocket was angled approximately 30° off of the vertical. Thus, both spin-plane and rocket-axis channels contained significant vertical and horizontal components.

There are also broad bands of VLF hiss, which could be auroral hiss. Banding in time/altitude is due to the rocket rotation. Within the cloud layer the hiss appears to have a lower frequency cutoff of about 8 kHz. The cloud layer appears to attenuate the hiss, since above it the signal is somewhat stronger overall, and the lower cutoff is around 4 kHz. Though this hiss is unrelated to the NLC layer, it supports the conclusion that the probes measured geophysical VLF signals. Pfaff et al. (2001) observed similar signals in the DROPPS campaign.

Figure 9 shows a power spectral density of the E_{15} electric field measured within the electron “bite-out” region from 85–87 km. There is a “knee” in the spectrum around 250 Hz, which corresponds to a spatial scale of about 4 m. The probe separation is about 0.65 m, so boom length effects should be marginal at those spatial scales. The spectrum of E_{24} has similar features. We expect structures to extend to these short scales, since radar reflects most strongly off features at the Bragg scale. Our spectrum is similar to spectra measured by Pfaff et al. (2001) on DROPPS.

4 Conclusions

We have presented high time-resolution E-field data in a NLC event. Similar to the DROPPS 1 rocket flight into a weak NLC, the MASS rocket flights found only weak DC electric field perturbations in the cloud. This finding contradicts theories which require large electric fields for the formation or maintenance of NLC conditions. The density profile of the heaviest aerosols measured by the MASS instrument had a similar shape to the skin-to-probe voltage. Such a strong similarity suggests that the skin-to-probe voltage was, at least within the cloud layer, perturbed by incident heavy charged aerosols. The probes measured a very sharp boundary at the lower edge of the NLC layer, in which the signal from the aft probes was delayed with respect to the forward probes caused by their 3 m separation. Thus, the structures seen in the skin-to-probe voltage are due to changes in the probe potential rather than the rocket skin potential. Similar potential shifts, in which the skin-to-probe voltage increased within the NLC event, were also seen in the DROPPS E-field data, and it is likely that those were likewise due to changes in the probe potential.

Spikes in the probe potential due to probe shadowing were observed in both MASS flights as well as both DROPPS flights. In DROPPS, the mechanisms controlling the am-

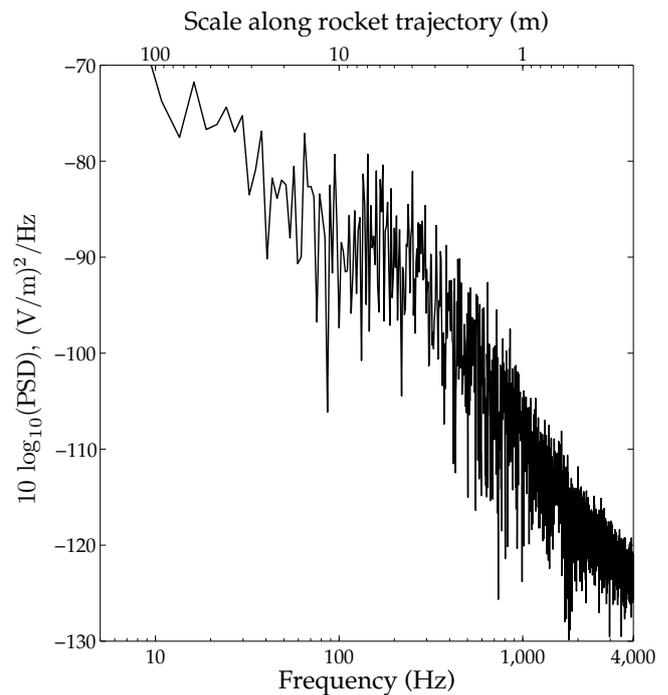


Fig. 9. Power spectral density (PSD) of E_{15} within the electron bite-out region. To remove probe shadowing artifacts from the data, spectra were calculated for the regions between the shadowing spikes. The E_{15} channel was used because the shadowing spikes occur half as often as in the E_{12} data. There is a “knee” in the spectrum around 250 kHz, which corresponds to a spatial scale of 4 m.

plitude of the spikes were unknown. With the first MASS flight, however, the sharpness of the NLC layer’s onset and the presence of the MASS instrument provided us with some unique calibration opportunities. Our calibration yielded results in rough agreement with those found in Croskey et al. (2001), despite using a completely different method. This was a non-trivial task, since the probes are designed to be used for difference measurements, whereas the calculation of conductivity largely relied on single-ended data.

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